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Application No.: 07/816,268

Please amend claims 1, 3, 5 and 7 as follows:

1. (Amended) An exhaust nozzle for a gas turbine engine capable of vectoring a flow of exhaust gases issuing therefrom at an angle to a centre line of the engine, the exhaust nozzle comprising a plurality of radially outer and radially inner flaps circumferentially disposed around the engine, the radially outer and radially inner flaps overlapping each other alternately to define a flow path through which the exhaust gases flow, means for moving the radially outer flaps asymmetrically about the engine centre line being provided, the radially inner flaps being urged into sealing contact with the radially outer flaps by the exhaust gas flowing through the nozzle, each of the radially inner flaps being divided into a plurality of triangular sections, so it is torsionally flexible so as to allow [them] it to twist along [their lengths so that] its length, whereby sealing contact is maintained with the radially outer flaps when the radially outer flaps are moved asymmetrically.

Claim 3, line 1, change "2" to --1--.

Claim 5, line 1, change "4" to --1--;

line 4, change "an" to --a--.

Claim 7, line 2, change "portions" to --sections--.

REMARKS

Upon entry of this Amendment, claims 1, 3 and 5-8 are pending. By the present Amendment, claims 2 and 4 are canceled and claims 1, 3, 5 and 7 are amended.

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